Numerical solution for solving procedure for 3D motions near libration points in the Circular Restricted Three Body Problem (CR3BP)

V. Christianto¹, & F. Smarandache²

¹ Malang Institute of Agriculture, Indonesia,

e-mail: victorchristianto@gmail.com

² Dept. Mathematics and Sciences, University of New Mexico, NM, USA, e-mail: smarand@unm.edu

Abstract

In a recent paper in *Astrophysics and Space Science Vol. 364* no. 11 (2019), S. Ershkov & D. Leschenko presented a new solving procedure for Euler-Poisson equations for solving momentum equations of the CR3BP near libration points for uniformly rotating planets having inclined orbits in the solar system with respect to the orbit of the Earth. The system of equations of the CR3BP has been explored with regard to the existence of an analytic way of presentation of the approximated solution in the vicinity of libration points. A new and elegant *ansatz* has been suggested in their publication whereby, in solving, the momentum equation is reduced to a system two coupled Riccati ODEs. In this paper, we presented a numerical solution of such coupled Riccati ODEs using Mathematica software package.

Keywords: Circular Restricted Three-Body Problem (CR3BP), Poisson equations, *Riccati* equation, *Jacobi* integral.

1. Introduction.

The equations of motion of the R3BP describe the dynamics of infinitesimal mass m under the action of gravitational forces affected by two celestial bodies of giant masses (in this problem M sun and m planet, m planet < M sun), which are rotating around

$$r_{H} = a_{p} \cdot \left(\frac{m_{planet}}{3M_{Sun}}\right)^{\frac{1}{3}}$$
(*)

their common centre of masses on Kepler's trajectories. The aforesaid infinitesimal mass m is supposed to be moving (as first approximation) inside the *restricted* region of space around the planet of mass m_{planet} or inside the so-called *Hill sphere* (where a_p is semimajor axis of the planet). [1]

It is worth to note that a large amount of previous and recent works concerning analytical development with respect to these equations exist. We can mention a few of them here.

Motivated by trajectory design applications, analytical work from the 1970s includes Farquhar and Kamel's third-order expansion of orbits in the vicinity of the Earth– Moon L2 libration point using a Poincar'e-Lindstedt method.5 During the same time period, Richardson and Cary developed a third-order approximation of quasi-periodic motion near the Sun–Earth L1 and L2 libration points via the method of multiple time scales. More recent semi-analytical work leverages computer algebra tools. G'omez et al. use a Poincar'eLindstedt method to generate high-order quasi-halo orbit expansions. Jorba and Masdemont investigate the dynamics around the libration points using center manifold reduction. These local methods offer a thorough view of the dynamics in the libration point vicinity, but they are limited by their regions of convergence. Specialized algebraic manipulators are often required as well, which can be an implementation obstacle. Still, the solutions have been successfully exploited for generating heteroclinic connections.[5]

In the 1980s Howell and Pernicka presented a numerical shooting approach for correcting approximate quasi-periodic orbits, though lacking control of orbit parameters. Gomez and Mondelo later developed a scheme for computing twodimensional quasi-periodic tori by using an invariant circle parameterized by Fourier coefficients of a stroboscopic map. Kolemen et al. use a similar approach except with multiple Poincare maps and directly parameterizing the invariant circle using states.1 The current approach uses similar concepts to these methods but with a more general formulation. We use a stroboscopic map, which requires less knowledge of the solution structure than a sequence of Poincare maps, and consider tori of arbitrary dimension. We use a grid of states on the invariant torus allowing us to avoid a convolution that is necessary when using a Fourier coefficient parameterization. In addition, we incorporate a different set of solution and continuation constraints that are broadly applicable. This allows the method to be used unchanged between families of solutions. An alternative approach presented by Schilder *et al.* computes a torus of a flow directly, and some of the constraints used have been adapted from this approach. The flow approach has been applied to the CR3BP; however, this requires dealing with a torus of dimension one larger than the torus of an associated map.[5]

We should especially emphasize the theory of orbits, which was developed in profound work [4] for the case of the Circular Restricted Problem of Three Bodies (CR3BP) (primaries M sun and m planet are rotating around their common centre of masses on *circular* orbits). According to [4], equations of motion of the infinitesimal mass m should be presented in the co-rotating frame of a Cartesian coordinate system $\vec{r} = \{x, y, z\}$ in case of CR3BP (*at given initial conditions*).

The suggested ansatz could be useful from a practical point of view in celestial mechanics (for example, to obtain a regular data of astrometric observations). Indeed, sometimes it is convenient for an observer to consider such the celestial dynamical system at a fixed angle to the plane of mutual circular rotation of the chosen primaries e.g. in the dynamics of planetary rotation in the Solar system.[1]

The most common example is Pluto's angle of orbit's inclination with respect to Earth's orbit (which is 17°9'); namely, *Pluto's* orbit is *inclined*, or tilted, circa 17.1 degrees from the *ecliptic* of *Earth's* orbit. Except Mercury's inclination of 7 degrees, all the other orbits of planets are closer to the orbit of Earth.[1]

2. Coupled Riccati ODEs.

According to the ansatz for solving *Poisson* equations, the system of Equations for three problem has *the analytical* way to present the general solution in regard to the time *t*:[1]

$$U = -\sigma \cdot \left(\frac{2a}{1 + (a^2 + b^2)}\right), \quad V = -\sigma \cdot \left(\frac{2b}{1 + (a^2 + b^2)}\right),$$
(1)
$$W = \sigma \cdot \left(\frac{1 - (a^2 + b^2)}{1 + (a^2 + b^2)}\right),$$

here $\sigma = \sigma(x, y, z)$ is some arbitrary (real) function, given by the initial conditions; the real-valued coefficients a(x,y,z, t), b(x,y,z, t) in (1) are solutions of the mutual system of two *Riccati* ordinary differential equations, ODEs in regard to the time t (here k = 2):

$$\begin{cases} a' = \left(\frac{k \cdot w_2}{2}\right) \cdot a^2 - (k \cdot w_1 \cdot b) \cdot a - \frac{k \cdot w_2}{2} (b^2 - 1) + (k \cdot w_3) \cdot b, \\ b' = -\left(\frac{k \cdot w_1}{2}\right) \cdot b^2 + (k \cdot w_2 \cdot a) \cdot b + \frac{k \cdot w_1}{2} \cdot (a^2 - 1) - (k \cdot w_3) \cdot a. \end{cases}$$
(2)

System (2) describes the evolution of function a in dependence on the function b in regard to the time t (and vice versa); we should especially note that the *Riccati* ODE has no analytical solution in general case.

3. <u>Numerical solution of equation (2)</u>

As an alternative approach to aforementioned solution as described in [1], we would like to prove that numerical solutions exist for equation (2). We will obtain the numerical solution using Mathematica software package.

 $\beta = k \cdot w_2 = 0.25$ $g = k \cdot w_1 = 0.15$ $ybar = k \cdot w_3 = 0.25$

 $\label{eq:param} param=\{\beta > 0.15, g > 0.15, y bar > 0.25, \gamma > 1.0, \delta > 0.03, Pstar > 0.025, iM > 0\}; \\ eqn1=P'[t]==(\beta/2)*P[t]^2-(g^*c[t])*P[t]-(\beta/2)*(c[t]^2-1)+y bar*c[t]/.param; \\ param=(\beta/2)*P[t]^2-(g^*c[t])*P[t]-(\beta/2)*(c[t]^2-1)+y bar*c[t]/.param; \\ param=(\beta/2)*P[t]^2-(g^*c[t])*P[t]-(g$

$$\begin{split} & eqn2=c'[t]==(-(g/2)^*c[t]^2+(\beta^*P[t])^*c[t]+(g/2)^*(P[t]^2-1)-\\ & ybar^*P[t])/.param;\\ & sol=Flatten[NDSolve[\{eqn1,eqn2, P[0]==-0.15,c[0]==0.7\},\{P,c\},\{t,-20,10\}]];\\ & plt1=ParametricPlot[\{P[t],c[t]\}/.sol,\{t, -7,7\}, Frame->True, FrameLabel-\\ \end{split}$$

>{Style["π",18],Style["c", 18]},AspectRatio->1]



sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-0.1,c[0]==0.7},{P, c},{t,-20,10}]]; plt2=ParametricPlot[{P[t],c[t]}/.sol,{t, -7,7}, Frame->True, FrameLabel->{Style["π",18],Style["c", 18]},AspectRatio->1]; sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-.07,c[0]==0.7},{P, c},{t,-30,15}]]

plt3=ParametricPlot[{P[t],c[t]}/.sol,{t, -10,10}, Frame->True, FrameLabel->{Style["π",18],Style["c", 18]},AspectRatio->1]; sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-.06,c[0]==0.7},{P, c},{t,-30,15}]] $\label{eq:plt4=ParametricPlot[{P[t],c[t]}/.sol, {t, -10,10}, Frame->True, FrameLabel->{Style["\pi,18],Style["c", 18]}, AspectRatio->1]; \\ sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-.05,c[0]==0.7}, {P, c}, {t, -30,15}]] \\ plt5=ParametricPlot[{P[t],c[t]}/.sol, {t, -10,10}, Frame->True, FrameLabel-} \\ \end{tabular}$

 $>{Style["\pi",18],Style["c", 18]},AspectRatio->1];$





Case 2:

 $\beta = k \cdot w_2 = 0.75$ $g = k \cdot w_1 = 0.35$ $ybar = k \cdot w_3 = 0.25$

 $param=\{\beta > 0.75,g > 0.35,ybar > 0.25,\gamma > 1.0,\delta > 0.03,Pstar > 0.025, iM > 0\};\\eqn1=P'[t]==(\beta/2)*P[t]^2-(g*c[t])*P[t]-(\beta/2)*(c[t]^2-1)+ybar*c[t]/.param;\\eqn2=c'[t]==(-(g/2)*c[t]^2+(\beta*P[t])*c[t]+(g/2)*(P[t]^2-1)-ybar*P[t])/.param;\\sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-0.15,c[0]==0.7},{P, c},{t,-20,10}]];\\plt1=ParametricPlot[{P[t],c[t]}/.sol,{t, -7,7}, Frame->True, FrameLabel->{Style["<math>\pi$ ",18],Style["c", 18]},AspectRatio->1]



sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-0.1,c[0]==0.7},{P, c},{t,-20,10}]]; plt2=ParametricPlot[{P[t],c[t]}/.sol,{t, -7,7}, Frame->True, FrameLabel->{Style["π",18],Style["c", 18]},AspectRatio->1]; sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-.07,c[0]==0.7},{P, c},{t,-30,15}]]

Show[{plt1, plt2, plt3, plt4, plt5}, AspectRatio \rightarrow 1, Axes \rightarrow False]



Case 3:

 $\beta = k \cdot w_2 = 0.15$ $g = k \cdot w_1 = 0.15$ $ybar = k \cdot w_3 = 0.01$ param={ β ->0.15,g->0.15,ybar->0.01, γ ->1.0, δ ->0.03,Pstar->0.025, iM->0};

```
eqn1=P'[t]==(β/2)*P[t]^2-(g*c[t])*P[t]-(β/2)*(c[t]^2-1)+ybar*c[t]/.param;
eqn2=c'[t]==(-(g/2)*c[t]^2+(β*P[t])*c[t]+(g/2)*(P[t]^2-1)-
ybar*P[t])/.param;
sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-0.15,c[0]==0.7},{P, c},{t,-20,10}]];
plt1=ParametricPlot[{P[t],c[t]}/.sol,{t, -7,7}, Frame->True, FrameLabel-
```

>{Style["π",18],Style["c", 18]},AspectRatio->1]



sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-0.1,c[0]==0.7},{P, c},{t,-20,10}]]; plt2=ParametricPlot[{P[t],c[t]}/.sol,{t, -7,7}, Frame->True, FrameLabel->{Style["π",18],Style["c", 18]},AspectRatio->1]; sol=Flatten[NDSolve[{eqn1,eqn2, P[0]==-.07,c[0]==0.7},{P, c},{t,-30,15}]]

 $> Style["\pi", 18], Style["c", 18], AspectRatio->1];$

Show[{plt1, plt2, plt3, plt4, plt5}, AspectRatio \rightarrow 1, Axes \rightarrow False]



The above shematic diagrams show numerical results of equation (2) in three different sets of parameters.

4. Discussion and conclusion.

As we can see in [1], the system of equations for the Circular Restricted Problem of Three Bodies (CR3BP) (which governs the dynamics of infinitesimal mass m under the action of gravitational forces affected by two mutually rotating celestial bodies of giant masses) has been proved to be very complicated to solve analytically in 3D case.

Indeed, numerical solutions of the mutual system (2) of two *Riccati* ordinary differential equations in regard to the time *t*, *can be found using Mathematica*. We also note that due to the special character of the solutions of *Riccati*-type ODEs, there is the possibility for sudden *jumping* in the magnitude of the solution at some time t_0 (see e.g. the solutions of *Riccati*-type for *Abel* ODE of 1-st order). In the physical sense, such jumping of *Riccati*-type solutions could be associated with the effect of a sudden acceleration/deceleration of the celestial body's velocity at a

definite moment of parametric time to.

5. Acknowledgements.

Authors are thankful to unknown esteemed reviewers with respect to the valuable efforts and advices which have improved structure of the article significantly.

Conflict of interest

Authors declare that there is no conflict of interests regarding publication of the article.

In this research, Victor Christianto is responsible for providing with the numerical calculations and graphical plots in Section 2 by means of advanced numerical methods as well as is responsible for numerical data of calculations.

Florentin Smarandache provides literature and proofread the entire manuscript. Both authors agreed with the results and conclusions of each other.

References:

- [1]. Ershkov S.V., Leshchenko, D. (2020). Solving procedure for 3D motions near libration points in CR3BP. Astrophys. Space. Sci. (2019) 364:1-7; DOI 10.1007/s10509-019-3692-z
- [2]. Arnold V. (1978). Mathematical Methods of Classical Mechanics. Springer, New York.
- [3]. Ershkov S.V., Shamin R.V. (2018). *The dynamics of asteroid rotation, governed by YORP effect: the kinematic ansatz.* Acta Astronautica, vol. 149, August 2018, pp. 47–54.
- [4]. Szebehely V. (1967). *Theory of Orbits. The Restricted Problem of Three Bodies.* Yale University, New Haven, Connecticut. Academic Press New-York and London.
- [5]. Zubin P. Olikara and Daniel J. Scheeres. NUMERICAL METHOD FOR COMPUTING QUASI-PERIODIC ORBITS AND THEIR STABILITY IN THE RESTRICTED THREE-BODY PROBLEM. IAA-AAS-DyCoSS1-08-10

- [6] R. C. Calleja · E. J. Doedel · A. R. Humphries · A. Lemus-Rodr'iguez · B. E. Oldeman. Boundary-Value Problem Formulations for Computing Invariant Manifolds and Connecting Orbits in the Circular Restricted Three Body Problem. Celest Mech Dyn Astr (2012) 114: 77. url: https://arxiv.org/abs/1111.0032v2
- [7] J.R. Iuliano. A SOLUTION TO THE CIRCULAR RESTRICTED N BODY PROBLEM IN PLANETARY SYSTEMS. A Thesis presented to the Faculty of California Polytechnic State University, San Luis Obispo, 2016.